

Swiss Confederation

Legal base:

Federal Office of Civil Aviation FOCA Safety Division - Aircraft

Lufttüchtigkeitsanweisung (LTA) Consigne de Navigabilité (CN) Direttive sulla Navigabilità (DN) Airworthiness Directive (AD)		FOCA AD HB-2020-013-E	
Inkraftsetzung Mise en vigueur Entrata in vigore Effective Date	ecember 2020	Pilatus – PC-9 FOCA TC/TCDS No: F 56-22	
Issue Date:	10 December 2020		
ATA Chapter:	ATA 25 – Equipment and Furnishings		
Subject:	Equipment and Furnishings – Ejection Seats – Harley-Type Buckle Assemblies on the Seat Harnesses – Inspection		
Foreign AD	Not applicable		
Supersedure / Revised AD(s):	Not applicable		
Type Certificate Holder's Name:	Pilatus Aircraft Ltd.		
Manufacturer(s):	Pilatus Aircraft Ltd., Hawker de Havilland Aerospace Pty. Ltd.		
Applicability:	Model PC-9, PC-9/A, PC-9/B and PC-9/F aeroplanes, all Manufacturers Serial Numbers (MSN).		
Definitions	For the purpose of this AD, the following definitions apply:		
	The Service Bulletin (SB)s: Pilatus Aircraft PC-9 SB 25-009. IrvinGQ SB IGQSB033.		
	The Special Information Leaflet: Martin-Baker Aircraft Company Special Information Leaflet No. 864		
	Affected part: Harley-type buckle assemblies installed on the seat harnesses of the front and rear ejection seats with incorrectly installed release bar retaining screws on a harness.		
	Serviceable part: Harley-type buckle assemblies installed on the seat harnesses of the front and rear ejection seats with correct installed release bar retaining screws on a harness shoulder strap buckle.		
	Affected aerople Any aeroplane h	ane: aving an affected part installed.	
Reason:	release bar retai	vas reported where in an in service event a missing ning screw on a Harley-type buckle assembly installed on der strap on an ejection seat was detected.	
		not corrected, could lead to loss of pilot restraint and s of aeroplane control or severe injuries to the crew.	
Base juridique:Art. 26, 51Base giuridica:Art. 26, 51	et 51A (Ordonnance sur la e 51A (Ordinanza concern	∙ die Luftüchtigkeit von Luftfahrzeugen: SR 748.215.1) navigabilité des aéronefs: SR 748.215.1) ente la navigabilità degli aeromobili: SR 748.215.1) e airworthiness of aircraft: SR 748.215.1)	

Art. 26, 51 e 51A (Ordinanza concernente la navigabilità degli aeromobili: SR 748.215.1) Art. 26, 51 and 51A (Ordinanza concernente la navigabilità degli aeromobili: SR 748.215.1)

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	To address this potential unsafe condition, Pilatus and IrvinGQ issued the SBs and Martin-Baker issued the Special Information Leaflet to provide inspection instructions.		
Required Action(s) and Compliance Time(s):	For the reason described above, this AD requires the inspection of the Harley-type buckle assemblies on the seat harnesses of the front and rear ejection seats, as defined in this AD, and prohibits (re-)installation of affected parts.		
	Required as indicated below, unless already accomplished:		
	 Inspection: (1) Before the next flight after the effective date of this AD, inspect the Harley-type buckle assemblies on the seat harnesses of the front and rear ejection seats in accordance with the instructions of section 3.B. of the Pilatus SB, the Martin-Baker Special Information Leaflet and section 2 of the IrvinGQ SB. 		
	 Corrective action(s): (2) If, during the inspection as required by paragraph (1) of this AD, discrepancies are detected, as identified in the applicable SBs and the Special Information Leaflet, before next flight, accomplish the applicable corrective action(s) in accordance with the instructions of the applicable SBs and the Special Information Leaflet. 		
	type buckle a	lation: ective date of this AD, it is allowed to install a Harley- assembly on the seat harnesses of the front and rear ts, provided it is a serviceable part, as defined in this AD.	
Ref. Publication(s):	Pilatus Aircraft PC-9 SB 25-009 original issue dated 27 November 2020.		
	Martin-Baker Aircraft Company Special Information Leaflet No. 864 original issue dated 25 November 2020.		
	IrvinGQ SB IGQSB033 original issue dated 20 November 2020.		
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.		
For further information contact:	The applicable manufacturer's documentation may be obtained directly from:		
	PILATUS AIRCRAFT LTD. Customer Technical Support P.O. Box 992 CH-6371 Stans, Switzerland		
	Tel.: +41 (0)41 619 3076 E-mail: <u>Techsupport@pilatus-aircraft.com</u>		
	For further information contact:		
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