

AIRWORTHINESS LIMITATIONS

1. Effectivity

Task effectivity is shown by the letter in the Code column of the Airworthiness Limitations table. The aircraft effectivity for each code letter is given in the Effectivity table below.

| EFFECTIVITY | | | |
|-------------|---|--|--|
| Code | Effectivity by Manufacturer's Serial Number | | |
| Р | ALL. PC-9M. | | |
| R1 | PC-9M MSN 617-633, 634-642, 643-654 | | |
| R5 | PC-9M MSN 605, 617-633, 634-642, 643-654 | | |
| Т | PC-9M MSN 617-633 | | |
| T2 | PC-9M MSN 605, 617-633 | | |

2. Airworthiness Limitations

| Task No. | Code | Task | Frequency |
|----------|------|---|---------------------------------------|
| 00006 | Р | Aircraft - General | 10000 flying hours or 20000 landings. |
| | | Aircraft life. | |
| 25204 | Р | Emergency Oxygen Cylinder (Ejection Seat) | 5 years. |
| | | Remove for overhaul (Ref. 25-10-07, Page Block 401). | |
| 28106 | Р | Aerobatic Fuel Tank | 6000 flying hours. |
| | | Replace clamp (Ref. CMM 02082, 28-10-01, Page Block 7001) | |
| 32108 | Р | Main Landing Gear Support Strut. | 6 months. |
| | | Examine. (Not Applicable if P/N 532.10.09.128 is installed), (Ref. 32-10-02, Page Block 601). | |
| 35301 | Т | Oxygen Cylinder | 15 years. |
| | | Replace (maximum life) (Ref. 35-10-02, Page Block 401). | |
| 35304 | Т | Oxygen Cylinder | 5 years. |
| | | Remove for hydrostatic test (Ref. 35-10-02, Page Block 401). | |



| Task No. | Code | Task | Frequency |
|----------|------|---|---|
| 53009 | R1 | Frame 3C to Longeron Connecting Bracket (Pre SB 53-005) | 2 years. |
| | | Examine (ejection seats, cockpit floor and side panels removed) (Ref. 53-10-03, Page Block 601). | |
| 53011 | Р | Longerons For aircraft older than 5 years, do a borescope | At the next annual inspection then every 2 years. |
| | | inspection. (Ref. 53-10-05, Page Block 601). | |
| 53602 | Р | Top Skin Slot at Frame 9 (not applicable if reinforcing doubler P/N: 553.30.09.394 has been installed). | First at 2600 flying hours, then every 1200 flying hours. |
| | | Examine (Ref. 53-10-02, Page Block 601). | |
| 53604 | Р | Fin Connecting Strap at Frame 9 | 5000 flying hours. |
| | | Replace (Ref. SRM 53-10-00, Page Block 201). | |
| 53608 | Р | Top Skin 60mm Access Hole | 5000 flying hours. |
| | | Examine (Applicable even if reinforcing doublers P/N 553.30.09.401 and 553.30.09.402 have been installed). | |
| 53609 | Р | Frame 8, 9 and 10 | 3600 flying hours. |
| | | For all aircraft with more than 5000 flying hours: Examine with a borescope. (Ref. 53-10-06, Page Block 601). | |
| 55101 | Р | Horizontal Stabilizer Installation Bolts | 5000 flying hours. |
| | | Replace (maximum life) (Ref. 55-10-00, Page Block 401). | |
| 55102 | | Deleted. | |
| 55304 | | Deleted. | |
| 55401 | R5 | Rudder Skin between Ribs 4 and 6 (only for rudders that do not have cracks). | 300 flying hours or 12 months. |
| | | Examine (Ref. 55-40-00, Page Block 601). | |
| 55402 | R5 | Rudder Skin between Ribs 4 and 6 (only for rudders that have cracks in limits). | Weekly. |
| | | Examine (Ref. 55-40-00, Page Block 601). | |
| 55505 | Р | Vertical and Horizontal Stabilizers Spar Flanges | 300 flying hours or 12 months. |
| | | Examine for corrosion (Ref. 55-05-00, Page Block 601). | |



| Task No. | Code | Task | Frequency |
|----------|------|--|---|
| 57003 | Р | Wing Installation Nuts and Bolts | 5000 flying hours. |
| | | Replace (maximum life) (Ref. 57-10-00, Page Block 401). | |
| 57011 | Р | Center Wing - Nose Skin | 300 flying hours. |
| | | Examine (Ref. 57-10-01, Page Block 601). | |
| 57013 | Р | Outer aileron bracket unless modification SB 57-012 implemented or material of bracket identified as AA2124-T851 during inspection SB 57-011. | 150 flying hours. |
| | | Do an Eddy Current inspection of the outer aileron bracket. (Ref. 57-21-00, Page Block 601). | |
| 57402 | R1 | Main Landing Gear Front Attachment Brackets, P/N: 111.34.07.105 and 111.34.07.106, Pre SB 57-008 or Post SB 57-008 but with cracks in limits (Ref. 57-10-08, Page Block 601) | Initial inspection at 3000 flying hours or 10 years in service, then every 12 months. |
| | | Do an impedance-plane eddy-current inspection (Ref. 57-10-08, Page Block 601). | |
| 57403 | R1 | Aft Dihedral Fittings | At 3000 flying hours or 10 years in service. |
| | | Do an impedance-plane eddy-current inspection and repair if applicable (Ref. 57-10-06, Page Block 601). | Sci vice. |
| 57405 | R1 | Aft Dihedral Fittings, P/N: 111.34.07.469 and 111.34.07.470, which have been inspected and found to have cracks in limits (Ref. 57-10-06, Page Block 601) | Every 1 year after initial inspection. |
| | | Do an impedance-plane eddy-current inspection (Ref. 57-10-06, Page Block 601). | |
| 57406 | R1 | Forward Dihedral Fittings | At 3000 flying hours or 10 years in service. |
| | | Do a radiographic inspection (Ref. 57-10-06, Page Block 601). | Scivice. |
| 57611 | Р | Diagonal Auxiliary Spar and Middle Rib Connection | First 1650 flying hours. Then every 160 flying hours. |
| | | Examine (Ref. 57-10-04, Page Block 601). | |
| 61107 | Р | Propeller Blades | 11500 flying hours. |
| | | Replace (maximum life) (Ref. Hartzell Manual 142). | |



| Task No. | Code | Task | Frequency |
|----------|------|---|---------------------|
| 71013 | Р | Engine Rotor Components | Ref. P&WC SB 13002. |
| | | Replace (maximum life). | |
| 71202 | Р | Engine Tubular Mounting Frame Nuts and Bolts | 5000 flying hours. |
| | | Replace (maximum life) (Ref. 71-20-01, Page Block 401). | |
| 71209 | Р | Engine Tubular Mounting Frame Attachment Fittings | 5 years. |
| | | Examine for corrosion and cracks. Results of the inspection are to be sent to Pilatus. (Ref. 71-20-01, Page Block 601). | |