

Swiss Confederation

Federal Office of Civil Aviation FOCA Safety Division - Aircraft

Lufttüchtigkeitsanweisung (LTA) Consigne de Navigabilité (CN) Direttive sulla Navigabilità (DN) Airworthiness Directive (AD)				FOCA AD HB-2011-006	
Inkraftsetzung Mise en vigueur Entrata in vigore Effective Date	10 August 2011			Pilatus – PC-9(M) FOCA TC/TCDS No: F 56-32	
Issue Date:		27 J	luly 2011		
ATA Chapter:		ATA 57 – Wings			
Subject:		Wings – LH & RH Outboard Aileron Bearing Brackets – Inspection			
Supersedure / Revised AD(s):		Not applicable			
Type Certificate Holder's Name:		Pilatus Aircraft Ltd.			
Manufacturer(s):		Pilatus Aircraft Ltd.			
Applicability:		Model PC-9(M) aircraft, Manufacturer Serial Numbers (MSN) 605 and MSN 617 through MSN 670 inclusive, without Service Bulletin 57-012 accomplished.			
Reason:		This Airworthiness Directive (AD) is prompted due to the discovery of cracks in the outboard aileron bearing brackets. The cracks are caused by stress corrosion. It is possible for stress corrosion cracks to occur in brackets initially made of aluminium alloy 2024-T351. Later in production, the material specification was changed to aluminium alloy 2124-T851 to decrease the risk of stress corrosion. The Part Number (P/N) of the brackets remained the same.			
		Such a condition, if left uncorrected, could lead to failure of the brackets and possible jammed or lost aileron in flight.			
		In order to correct and control the situation, this AD requires a one-time check to identify the material specification and inspect those affected brackets that are made of aluminium alloy 2024-T351. Any brackets found to be cracked are to be replaced prior to further flight.			
Required Action and Compliance	(s)	Req	uired as indicat	ed below, unless already accomplished:	
Time(s):		(1)	tivity test to ide P/N 557.22.09	h after the effective date of this AD, perform a conduc- entify the material specification of the brackets (LH, 0.281 & RH, P/N 557.22.09.282) as required by para- of PILATUS PC-9(M) Service Bulletin No. 57-011.	
		(2)	made of alumi perform the in	onductivity test required by § (1) of this AD, brackets inium alloy 2024-T351 are found, prior to further flight, spection in accordance with (§) 3.C. of PILATUS PC- Bulletin No. 57-011.	

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	 Note 1: The Aircraft Maintenance Manual (AMM) will be updated to include a repeat inspection of all brackets made of aluminium alloy 2024-T351. (3) If during the inspection required by § (2) of this AD, cracks are found in the LH or RH brackets, prior to further flight, replace the brackets in accordance with PILATUS PC-9(M) Service Bulletin No. 57-012. 				
Ref. Publication(s):	PILATUS PC-9(M) Service Bulletin No. 57-011, initial issue. PILATUS PC-9(M) Service Bulletin No. 57-012, initial issue.				
	The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.				
For further information contact:	The applicable manufacturer's documentation may be obtained directly from:				
	PILATUS AIRCRAFT LTD. Customer Technical Support (MCC) P.O. Box 992 CH-6371 Stans, Switzerland				
	Tel.: +41 (0)41 619 32 22 Fax: +41 (0)41 619 67 73 E-mail: <u>Techsupport@pilatus-aircraft.com</u>				
	For further information contact:				
	FEDERAL OFFICE OF CIVIL AVIATION (FOCA) Design and Production (STEH) CH-3003 Bern, Switzerland				
	Fax: +41 (0) 31 322 59 18 E-mail: <u>airdir@bazl.admin.ch</u>				