

Swiss Confederation

Federal Office of Civil Aviation FOCA Safety Division - Aircraft

Lufttüchtigkeitsanweisung (LTA) Consigne de Navigabilité (CN) Direttive sulla Navigabilità (DN) Airworthiness Directive (AD)			FOCA AD HB-2009-010
Inkraftsetzung Mise en viguer Entrata in vigore Effective Date		ptember 2009	Pilatus – PC-9(M) FOCA TC/TCDS No: F 56-32
Issue Date:		02 September 2009	
ATA Chapter:		ATA 46 – Information Systems	
Subject:		Information Systems – Primary Flight Display – Replacement	
Supersedure / Revised AD(s):		Not applicable	
Type Certificate Holder's Name:		Pilatus Aircraft Ltd.	
Manufacturer(s):		Pilatus Aircraft Ltd.	
Applicability:		Model PC-9(M) aircraft, Manufacturer Serial Numbers (MSN) 655 through MSN 670 inclusive.	
Reason:		This Airworthiness Directive (AD) is prompted by reported incorrect flap position indication on the Primary Flight Display (PFD). Investigation revealed that the microswitch on the flap actuator, that controls take-off position and indication, had a broken spring. At the moment, the PFD has a priority logic implemented for a multiple flap input, which can cause, together with a broken microswitch, an incorrect flap position indication.	
		Such a condition, if left uncorrected, could lead to incorrect flap position and indication which may be undetected by the pilot. This can lead to:	
		 an unexpected increase of the required take-off length. an unexpected increase of stall speed with potential loss of control of the aircraft. 	
		In order to correct and control the situation, this AD requires a PFD soft- ware upgrade which implements a new logic. The new logic will be that if two or more flap inputs to the PFD are active, the flap indication on the PFD will be displayed in red.	
Required actions(s(s)	Required as indicated below, unless already accomplished:	
and Compliance Time(s):		units Part Nui accomplishme	ths after the effective date of this AD, remove the PFD mber (P/N) 975.96.32.722 in accordance with the ent instructions of paragraph (§) 3.A.(4) of PILATUS rice Bulletin (SB) No. 46-001.
		,	dified PFD units P/N 975.96.32.731 in accordance mplishment instructions of § 3.B. and C. of PILATUS No. 46-001.

After the effective date of this AD, no person shall install a PFD unit 3) P/N 975.96.32.722 on any post SB No. 46-001 aircraft as defined in the applicability paragraph. Ref. Publication(s): PILATUS PC-9(M) Service Bulletin No. 46-001, initial issue. The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD. For further information The applicable manufacturer's documentation may be obtained directly contact: from: PILATUS AIRCRAFT LTD. Customer Liaison Manager CH-6371 Stans, Switzerland Tel.: +41 (0)41 619 62 26 Fax: +41 (0)41 619 61 70 E-mail: snolan@pilatus-aircraft.com For further information contact: FEDERAL OFFICE OF CIVIL AVIATION (FOCA) Design and Production (STEH) CH-3003 Bern, Switzerland

Fax: +41 (0) 31 322 59 18 E-mail: <u>airdir@bazl.admin.ch</u>