

Swiss Confederation

Legal base:

Federal Office of Civil Aviation FOCA Safety Division - Aircraft

Lufttüchtigkeitsanweisung (LTA) Consigne de Navigabilité (CN) Direttive sulla Navigabilità (DN) Airworthiness Directive (AD)		FOCA AD HB-2019-002	
Inkraftsetzung Mise en vigueur Entrata in vigore Effective Date	11 February 2019	Pilatus – PC-7 FOCA TC/TCDS No: F 56-20	
Issue Date:	28 January 20	28 January 2019	
ATA Chapter:	ATA 24 – Elec	ATA 24 – Electrical Power	
Subject:	Electrical Pow	Electrical Power – Electrical Wires below Panel F5 – Inspection	
Supersedure / Re AD(s):	vised None		
Type Certificate Holder's Name:	Pilatus Aircraft	Pilatus Aircraft Ltd.	
Manufacturer(s):	Pilatus Aircraft	Pilatus Aircraft Ltd.	
Applicability:		Model PC-7 aircraft, Manufacturers Serial Numbers (MSN) 101 through MSN 537, MSN 548 through MSN 609 and MSN 613 through MSN 618.	
Definitions:	For the purpos	For the purpose of this AD, the following definitions apply:	
		The Service Bulletin (SB): Pilatus Aircraft Ltd. PC-7 SB No. 24-008 Revision 1.	
		Affected part: Electrical wires below panel F5.	
Reason:		During a scheduled inspection several chafed and burned wires located under panel F5 were found.	
		This condition, if not detected and corrected, could lead to smoke/fume in the cockpit and in a possible in-flight fire.	
		To address this potential unsafe condition, Pilatus Aircraft Ltd. issued the SB to provide applicable inspection instructions.	
	a visual inspec	For the reason described above, this Airworthiness Directive (AD) requires a visual inspection of the wires below the panel F5 and, depending on findings, accomplishment of applicable corrective action(s).	
Required Action(s	s) Required as in	Required as indicated below, unless already accomplished:	
and Compliance Time(s):	Within the next	Within the next 120 days after the effective date of this AD:	
	of chafing Units (LR (ECS) ho	on: a visual inspection of the wires below the panel F5 for signs g and contact with the casings of the Line Replacement Us), panel edges and/or Environmental Control System ses for signs of chafed and burned wires in accordance with ctions of Section 3.B. of the SB.	

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	 Corrective Action(s): (2) If, during the inspection as required by paragraph (1) of this AD, any signs of chafed and burned wires below the panel F5 are found, as defined in the SB, before next flight, replace the wires in accordance with the instructions of Section 2. and 3.B. of the SB. 	
Ref. Publication(s):	Pilatus Aircraft Ltd. PC-7 SB No. 24-008 Revision 1 dated 20 December 2018.	
	e use of later approved revisions of this document is acceptable for npliance with the requirements of this AD.	
For further information contact:	 The applicable manufacturer's documentation may be obtained directly from: PILATUS AIRCRAFT LTD. Customer Technical Support (MCC) P.O. Box 992 CH-6371 Stans, Switzerland 	
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