EASA AD No: 2008-0164

EASA

AIRWORTHINESS DIRECTIVE



AD No.: 2008-0164

Date: 27 August 2008

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

tate of Registry [EC 216/2006	3, Article 14(4) exemption].	
Type Approval Holder's Name : Pilatus Aircraft Ltd.		Type/Model designation(s): PC-6
TCDS Number :	FOCA Switzerland F 56-10	
Foreign AD :	Not applicable	
Supersedure: None		
ATA 32	Landing Gear - Mov	rement of Brake Pedals- Check/Modification
Manufacturer(s):		Fairchild Republic Company (formerly known as irchild Heli Porter and Fairchild-Hiller Corporation).
Applicability:	PC-6 aircraft, all model	s, all serial numbers
Reason:		ective (AD) is prompted by a potential problem with se pedals of some PC-6 series aircraft.
	insufficient clearance b such conditions, it is po with the fluid from the r effectiveness. Mostly d	ke pedals could be prevented because of an etween the rudder bar lugs on a few aircraft. In essible that the master brake cylinder is not re-filled eservoir, which can lead to a degradation of brake uring landing, this can lead to difficulties with the e aircraft on ground and could cause a runway
	brake pedals for full an	above, the present AD mandates a check of the d free movement and, if any damage is found, the se pedals to restore their freedom.
Effective Date:	10 September 2008	
Required Action(s) and Compliance Time(s):	(1) Within 100 Flight H	ours or 12 months after the effective date of this urs first, check the brake pedals for full and free

EASA Form 110 Page 1/2

	movement in accordance with accomplishment instructions of PILATUS PC-6 Service Bulletin No. 32-002 Revision 2.	
	If any stiffness or limited movement of a brake pedal is found, before next flight, perform the corrective actions in accordance with the paragraph 3.C. of the accomplishment instructions of PILATUS PC-6 Service Bulletin No. 32-002 Revision 2.	
	(2) After the effective date of this AD, no person shall install any pilot and co-pilot rudder bars and brake pedal assembly Part Number (P/N) 6232.0011.00, P/N 6232.0255.52, P/N 116.358.06.050 and P/N 116.358.06.053 on any PC-6 aircraft, unless they have been previously inspected in accordance with paragraph 4. of the accomplishment instructions of PILATUS PC-6 Service Bulletin No. 32-002 Revision 2.	
Ref. Publications:	PILATUS PC-6 Service Bulletin No. 32-002 Revision 2	
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.	
Remarks :	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD. This AD was posted on 31 July 2008 as PAD 08-092 for consultation until 21 August 2008. The Comment Response Document can be found at http://ad.easa.europa.eu. Enquiries regarding this AD should be referred to the Airworthiness Directives, Safety Management & Research Section, Certification Directorate, EASA. E-mail ADs@easa.europa.eu For any questions concerning the technical content of the requirements in this AD, please contact: PILATUS AIRCRAFT LTD., Customer Liaison Manager, CH-6371 STANS, Switzerland Tel.: +41 (0)41 619 65 80 Fax: +41 (0)41 619 65 76 E-mail: fodermatt@pilatus-aircraft.com 	

