EASA AD No.: 2014-0181

EASA

AIRWORTHINESS DIRECTIVE

AD No.: 2014-0181

Date: 31 July 2014

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation.

This AD is issued in accordance with EU 748/2012, Part 21.A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption].

Design Approval Holder's Name: PILATUS AIRCRAFT Ltd.		Type/Model designation(s): PC-6 aeroplanes
TCDS Number:	Switzerland No. F 56-10	To a desire particular to the second
Foreign AD:	Not applicable	
Supersedure:	This AD supersedes EASA AD	2012-0268 dated 19 December 2012.
ATA 05	Time Limits / Maintenan Section – Amendment /	ce Checks – Airworthiness Limitations Implementation
Manufacturer(s):	Pilatus Aircraft Ltd. and Fairchild Republic Company (formerly Fairchild Industries, Fairchild Heli Porter and Fairchild-Hiller Corporation).	
Applicability:	PC-6 aeroplanes, all models, all manufacturer serial numbers.	
Reason:	The maintenance instructions and airworthiness limitations applicable to the Structure and Components of PC-6 aeroplanes are specified in the Aircraft Maintenance Manual (AMM) under Chapter 4 or in the Airworthiness Limitations Document (ALS), depending on aeroplane model.	
	mandatory actions for contin	n the ALS document have been identified as ued airworthiness and failure to comply with thes ould potentially lead to an unsafe condition.
	Document Number 01975 is aeroplanes and PC-6 ALS, [recently issued PC-6 AMM, Chapter 04-00-00, sue 19 for PC-6 B2-H2 and PC-6 B2-H4 Document Number 02334 issue 4 for all other PC orate new life limits for the Fire Extinguisher.
	2012-0268, which is superse	ove, this AD retains the requirements of EASA Aleded, and requires implementation of the new and/or airworthiness limitations.
Effective Date:	14 August 2014	

EASA AD No.: 2014-0181

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

- (1) From the effective date of this AD, accomplish the following actions, as detailed in Chapter 4 of the AMM report, or in the ALS document, as specified in Table 1 of this AD, as applicable to aeroplane model:
 - (1.1) Replace each component before exceeding the applicable life limit, and
 - (1.2) Within the thresholds and intervals, accomplish all applicable maintenance tasks.

Table 1 – Applicable AMM Report / ALS Document

Aeroplane Models	AMM Report / ALS Document
PC-6 B2-H2 and PC-6 B2-H4	Chapter 04-00-00 of AMM Document Number 01975, issue 19 dated 31 May 2014
All other PC-6 models	ALS Document Number 02334, issue 4 dated 31 May 2014

(2) In case of finding discrepancies (as defined in Chapter 4 of the AMM report or in the ALS document specified in Table 1 of this AD) during accomplishment of any task as required by paragraph (1) of this AD, within the compliance time specified in Chapter 4 of the relevant AMM report or in the ALS document, accomplish the applicable maintenance procedures for corrective actions in accordance with the approved maintenance documentation.

If no compliance time is identified in Chapter 4 of the relevant AMM report or in the ALS document, accomplish the applicable corrective actions before next flight.

If a detected discrepancy is not identified in Chapter 4 of the relevant AMM report or in the ALS document, before next flight, contact Pilatus for approved instructions and accomplish those instructions accordingly.

- (3) Compliance with the requirements of paragraphs (1) and (2) of this AD can be demonstrated by:
 - (3.1) Revising, as follows, the approved Aircraft Maintenance Programme and standard practices on the basis of which the operator or the owner ensures the continuing airworthiness of each operated aeroplane:

Incorporate all applicable maintenance instructions and associated airworthiness limitations specified in Chapter 04-00-00 of AMM Document Number 01975 issue 19 or in the ALS Document Number 02334 issue 4, as applicable to aeroplane model,

and

(3.2) Complying with the approved Aircraft Maintenance Programme described in paragraph (3.1) of this AD.

Ref. Publications:

Pilatus PC-6 ALS Document Number 02334, issue 4 dated 31 May 2014.

Pilatus PC-6 AMM, Chapter 04-00-00, Document Number 01975, issue 19 dated 31 May 2014.

The use of later approved revisions of these documents is acceptable for compliance with the requirements of this AD.

EASA AD No.: 2014-0181

Remarks:

- 1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
- 2. This AD was posted on 01 July 2014 as PAD 14-108 for consultation until 29 July 2014. No comments were received during the consultation period.
- 3. Enquiries regarding this AD should be referred to the Safety Information Section, Executive Directorate, EASA. E-mail: ADs@easa.europa.eu.
- 4. For any question concerning the technical content of the requirements in this AD, please contact:

PILATUS AIRCRAFT LTD., Customer Liaison Manager, CH-6371 STANS, Switzerland.

Tel.: +41 (0)41 619 65 80 Fax: +41 (0)41 619 65 76.

E-mail: fodermatt@pilatus-aircraft.com.

