Bundesamt für Zivilluftfahrt (BAZL) Office fédéral de l'aviation civile (OFAC) Ufficio federale dell'aviazione civile (UFAC) Federal Office for Civil Aviation (FOCA)

Lufttüchtigkeitsanweisung Consigne de navigabilité Prescrizione di aeronavigabilità Airworthiness directive

HB 96-335

Inkraftsetzung Mise en vigueur Entrata in vigore Effective Date

31.07.1996

SUPERSEDED BY FOCA AD: HB 97-002

Betroffene Muster - Types concernés - Applicabilità - Models affected

PILATUS AIRCRAFT LTD. - All PC-12 aircraft, up to and including serial number 140.

Anlass/Massnahmen - Objet/Mesures - Oggetto/Provvedimenti - Subject/Action

AUTOMATIC DISCONNECT CIRCUIT FOR THE NON-ESSENTIAL BUS - Rewiring

NAVIGATION CONVERTER (KN 40)

- Inspection for correct wiring.
- Rectification if necessary.

Fristen - Délais - Scadenza - Compliance

Required before December 31, 1996 in accordance with the Accomplishment Instructions of Pilatus Service Bulletin PC-12 No.: 24-003. dated June 11.1996.

Herkunft - Provenance - Provenienza - Origin

FOCA AD

Bezugnahme - Référence - Riferimento - Reference

Pilatus PC-12 Service Bulletin No.: 24-003. dated June 11.1996.

The applicable Service Bulletins may be obtained directly from:

PILATUS AIRCRAFT LTD
MARKETING SUPPORT DEPARTMENT

CH-6370 STANS,

Tel No: +41 41 6196 233

SWITZERLAND

Fax No: +41 41 6103 351

Bemerkungen - Observations - Osservazioni - Remarks

For detailed informations refer to page 2

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HB 96-335 PILATUS AIRCRAFT LTD.

Applicability

PILATUS AIRCRAFT LTD.- All PC-12 aircraft, up to and including serial number 140.

Reason

- 1. When the N ESNTL switch is set to AUTO, the NON ESSENTIAL BUS must automatically disconnect from the power distribution system when the BAT/GEN 1 bus tie circuit breaker (CB210) opens. These wiring changes eliminate alternative grounding circuits for N ESNTL BUS relay (K331) which could cause the NON ESSENTIAL BUS to remain connected.
- 2. During production of a PC-12, an instance was discovered where the power lines of the Navigation Converter (KN 40) was incorrectly wired. In the instance of a power loss on Avionic Bus 1 and Avionic Bus 2 loss of remaining VOR/LOC information would be experienced.

Actions

Since the unsafe condition described has been identified that is likely to exist or develop on other aircraft of the same type design, this Airworthiness Directive requires rewiring and testing of the Automatic Disconnect Circuit for the Non Essential Bus and an inspection of the Navigation Converter (KN 40) for correct wiring.

Compliance

Required before December 31, 1996 in accordance with the Accomplishment Instructions of Pilatus Service Bulletin PC-12 No.: 24-003. dated June 11.1996.

FOR FURTHER INFORMATION CONTACT:

Federal Office for Civil Aviation (FOCA), Type Certification Section (FM), Maulbeerstrasse 9. CH-3003 Berne, Switzerland./ Facsimile (+41) 31 325 80 48.