EASA EMERGENCY AIRWORTHINESS DIRECTIVE AD No.: 2009-0028-E Date: 11 February 2009 Note: This emergency Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EC) No 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation This AD is issued in accordance with EC 1702/2003, Part 21A.3B. In accordance with EC 2042/2003 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of that Airworthiness Directive unless otherwise specified by the Agency [EC 2042/2003 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [EC 216/2008, Article 14(4) exemption]. Type Approval Holder's Name : Type/M del designation(s) : PILATUS AIRCRAFT Ltd. eries airplanes TCDS Number : EASA.A.089 Foreign AD : Not applicable Supersedure : None vigation - Incoldect Information of the Air Data, Attitude and Reference System (ADAHRS) - POH Temporary **ATA 34** Headin Revision Manufacturer Aircraft Lto Ph ato Model PC-12 47E airplanes with Manufacturer Serial Number (MSN) 545 Applicability: and from MSN 100 and up. Field ep rts bare indicated that during take-off with light turbulences or after Reason: rapid foll heading attitude changes soon after take-off, it is possible that both ary Flight Display) indicate a roll attitude offset of up to 10 degrees PFDs (Pr in the same direction. This condition has been reported to correct itself after several minutes of un-accelerated flight with levelled wings and no sideslip. This situation, if not corrected, could result in an undesired bank angle which constitutes an unsafe condition. For the reason described above, this Airworthiness Directive (AD) mandates as an interim measure a revision of the operational procedures to be inserted into the Pilot's Operating Handbook (POH). Effective Date: 13 February 2009

	(2) Insert the following information in Section 3A – Abnormal Procedures:
	When a discrepancy is observed between the PFD and the Electronic Standby Instrument System (ESIS), the aircraft should be flown with Flight Director (FD) and Auto Pilot (AP) engaged until the error corrects itself. The ESIS should be used as the primary means of attitude reference as long as the problem exists. If the problem does not disappear after 15 minutes, transition to VMC conditions for the rest of the flight is required.
	(3) Insert the following information in Section 4 – Normal Procedures:
	The FD shall be used for take-off in either heading or navigation mode as appropriate and with the heading or navigation properly set.
	The pilot shall include the ESIS in the normal scan.
	 (4) Incorporation of the Temporary Revision No. 9 or a copy of this AD into the Pilatus PC-12/47E POH satisfies the requirements of paragraphs (1), (2) and (3) of this AD.
Ref. Publications:	Temporary Revision No. 9 to 20-12/47E Pilot's Operating Handbook Report No.02277.
	The use of later approved revisions of this document is acceptable for compliance with the requirements of this AD.
Remarks :	 If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
	The safety assessment has requested not to implement the full consultation process and an immediate publication and notification.
	3. Exquines regarding this (<i>D</i> should be referred to the Airworthiness Directives, Safe Victoragement & Research Section, Certification Directorate, EAS, . E-mail: <u>A ISUes Deuropa.eu</u>
	4 For any rue tions concerning the technical content of the requirements in this AD, please contact: PLATES AIRCRAFT LTL, eastomer Support Manager,
5	CH-6372 STANS, Switzerlund 1915 +41 (0)41 o 9 62 08 Fax: +41 (0)41 619 73 11 e-mail: <u>SupportPO 2@pilatus-aircraft.com</u>